

# Ard- $\mu$ -copter: A Simple Open Source Quadcopter Platform

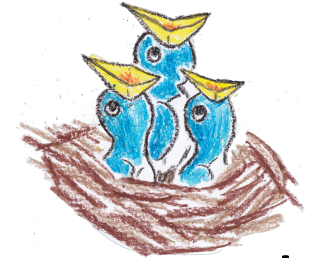
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The Embedded Systems and CPS Lab, Dept. of Computing  
The Hong Kong Polytechnic Univ.

December, 2015

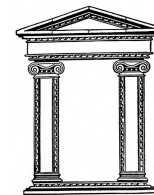




Conclusion



Demand



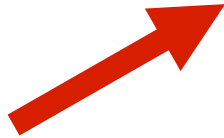
Architecture



Control Strategy



Evaluation





Demand: increasing usage of UAV (quadcopter) in various domains, and as perfect CPS testbed





Demand: increasing usage of UAV (quadcopter) in various domains, and as perfect CPS testbed

Photo/video taking





Demand: increasing usage of UAV (quadcopter) in various domains, and as perfect CPS testbed

Photo/video taking

Virtual reality gaming





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Photo/video taking

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Movie industry



Demand: increasing usage of UAV (quadcopter) in various domains, and as perfect CPS testbed

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Movie industry

Surveillance



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Movie industry

Transportation

Surveillance



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Photo/video taking

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Construction site monitoring

Movie industry

Transportation

Surveillance



Demand: increasing usage of UAV (quadcopter) in various domains, and as perfect CPS testbed





Demand: UAV (quadcopter) is a potential perfect CPS testbed





# Demand: UAV (quadcopter) is a potential perfect CPS testbed

Embedded systems





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Embedded systems

Real-time systems





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Embedded systems

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Control theory





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Aerodynamics





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Control theory

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Mechanics





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Robotics/SLAM/Pathplanning



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Computer vision

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Control theory

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Robotics/SLAM/Pathplanning

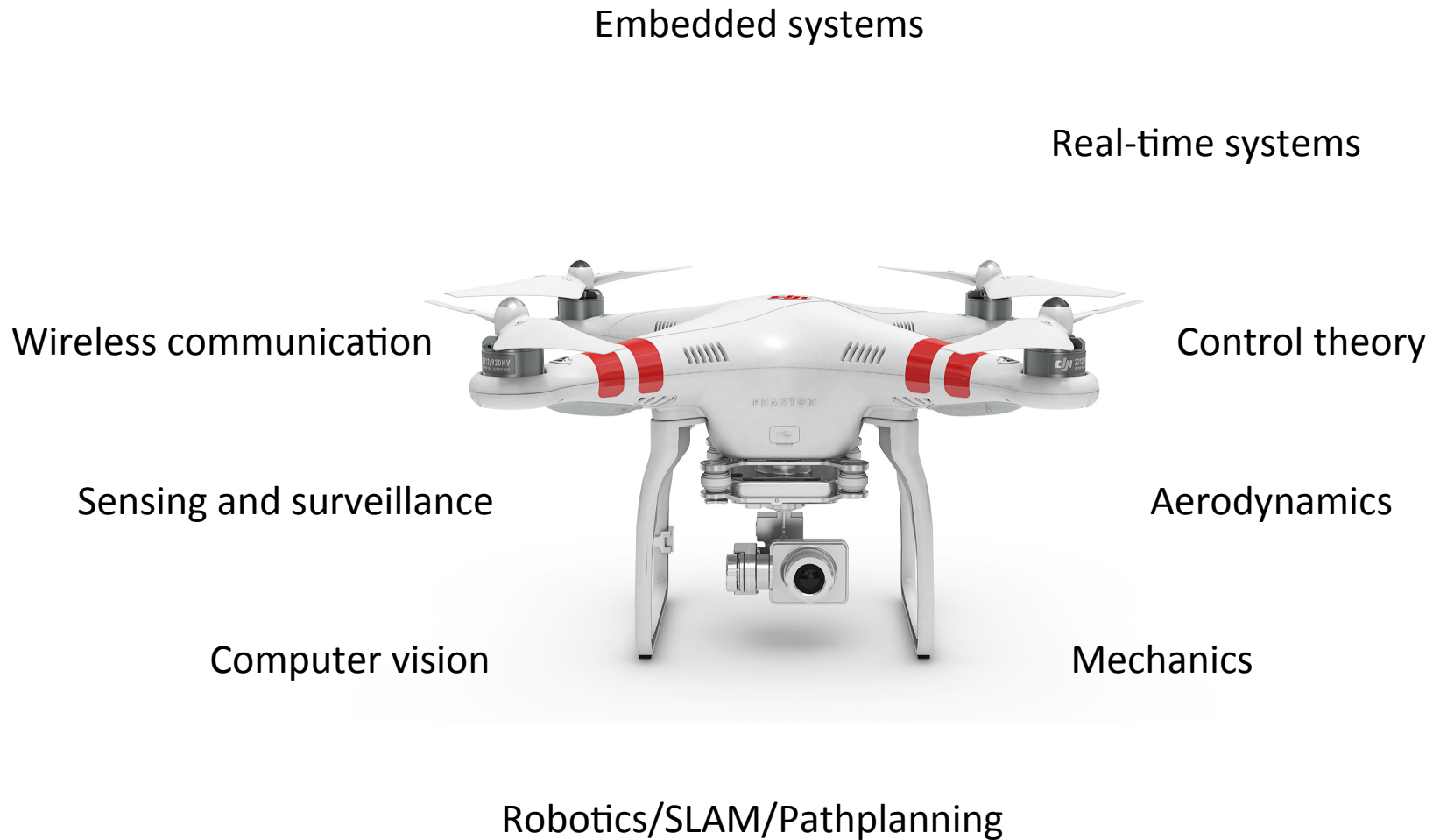
Sensing and surveillance

Computer vision



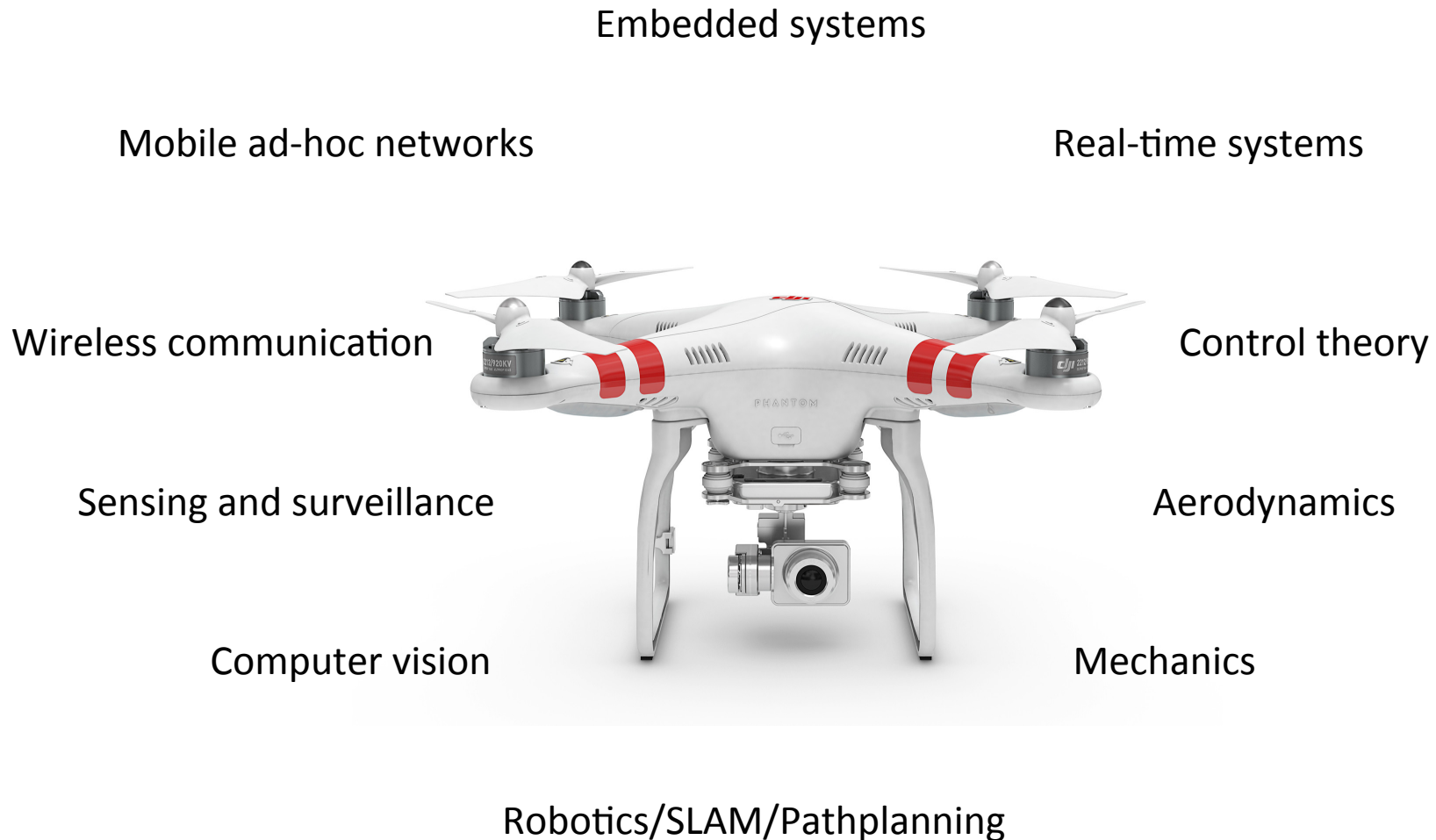


# Demand: UAV (quadcopter) is a potential perfect CPS testbed





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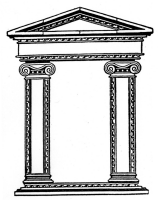




# From potential to actual perfect CPS test bed: open source, light weight, good documentation

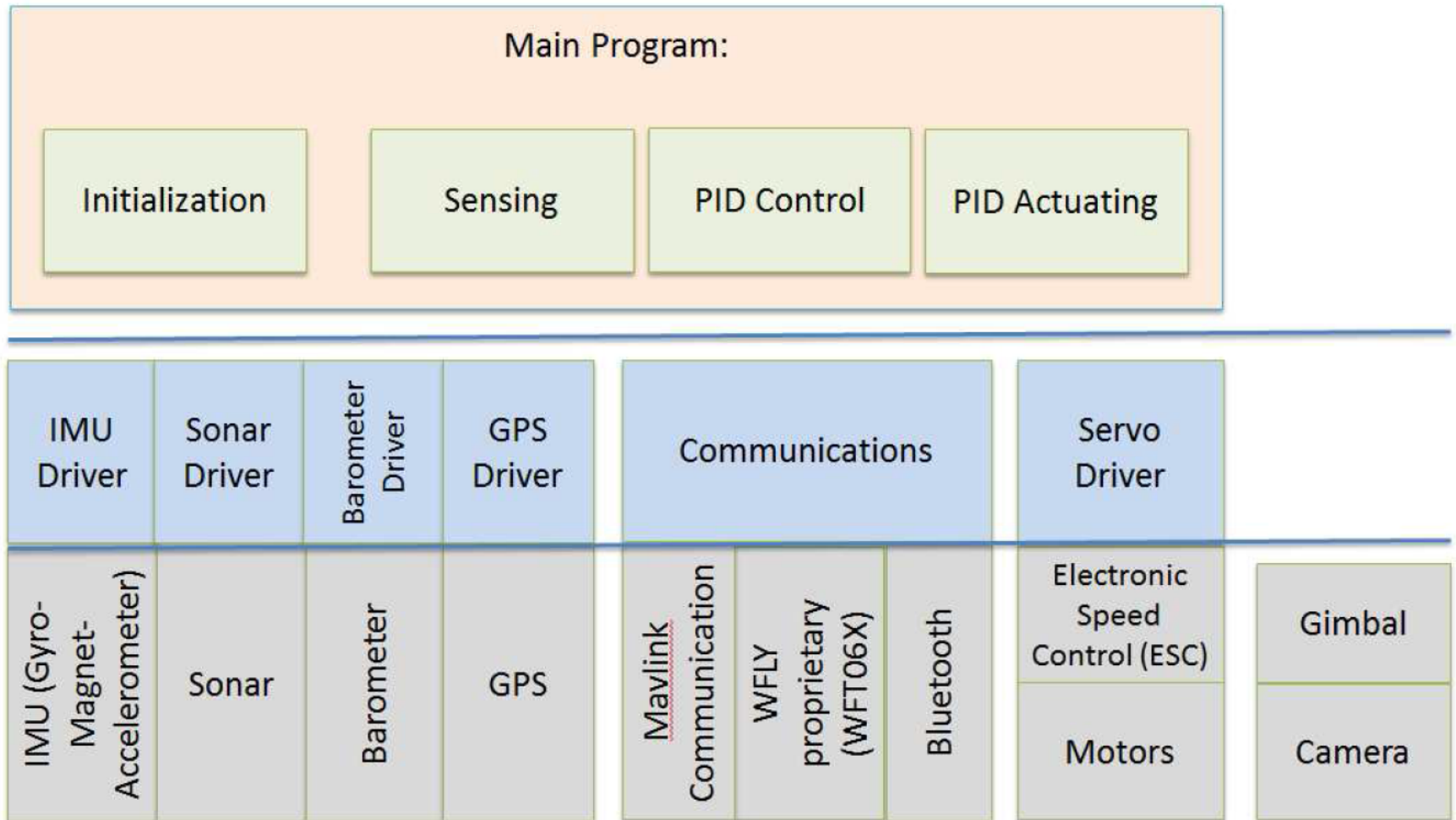


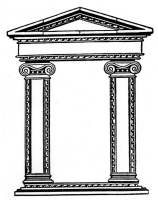
| Product                                                                       | DJI | MicroPilot | Ardupilot                 | Ard- $\mu$ -copter         |
|-------------------------------------------------------------------------------|-----|------------|---------------------------|----------------------------|
| Open upper application layer<br>(coarse grain control)                        | Yes | Yes        | Yes                       | Yes                        |
| Open source lower application layer<br>(fine grain control, attitude control) | No  | No         | Yes (20MB<br>source code) | Yes (135KB<br>source code) |
| Open source OS/driver layer                                                   | No  | No         | Yes (1MB<br>source code)  | Yes (1MB<br>source code)   |
| Open source hardware                                                          | No  | No         | Yes                       | Yes                        |



# Ard- $\mu$ -copter architecture

## Quadcopter Onboard System Architecture





# Ard- $\mu$ -copter architecture

## Quadcopter Onboard System Architecture

Main Program:

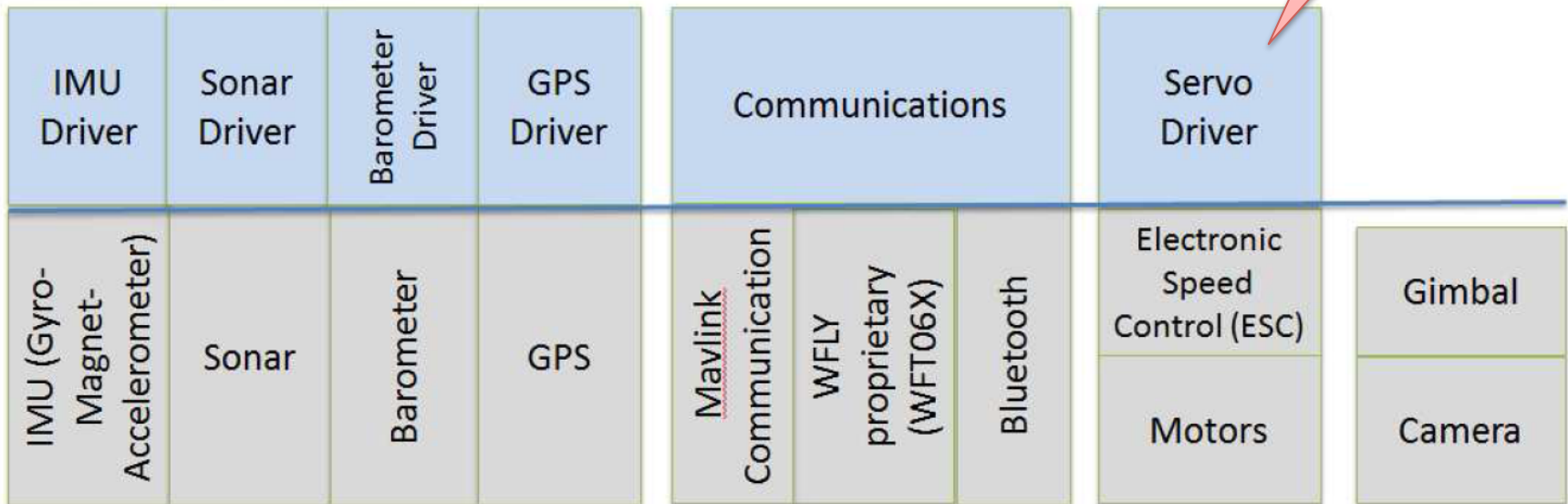
Initialization

Sensing

PID Control

PID Actuating

Reuses  
Ardupilot  
driver library  
(1MB source  
code)



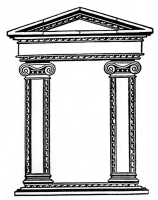


1. Fine grain attitude control (pitch-roll-yaw angles)
2. Interface for coarse grain x-y-z location control

### Main Program:

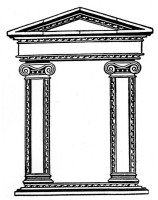
## PID Actuating

[illegible]



Platform architecture: hardware layer uses the open source ArduPilot Mega 2.5.2 (aka APM 2.5.2) board

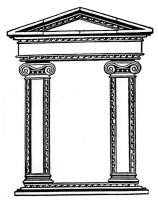
|              |                  |
|--------------|------------------|
| MCU          | ATmega 2560      |
| IMU          | MPU-6050         |
| Magnetometer | HMC5883          |
| GPS          | UBLOX-6H         |
| Barometer    | MS5611           |
| Sonar        | XL-MaxSonar EZ4  |
| Wireless     | Usmile Mini 433M |



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AVR architecture,  
8bit, 16MHz,  
86 input/output pins,  
AD/DA,  
UART/USART, SPI,  
I2C, EBI/EMI  
256MB Flash,  
4KB EEPROM, 8K RAM,

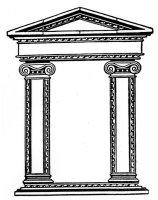


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3-axis gyro,  
3-axis accelerometer,  
16bit data,  
up to 400Hz sampling rate,  
SPI comm interface

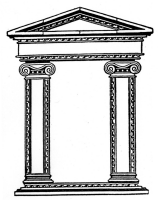
Additional Digital Motion  
Processing (DMP) engine to  
synthesize angle readings



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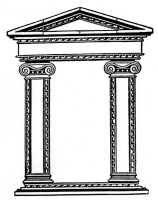
Range: 1.3~8 Gauss  
Package size: 1.8x1.3cm  
IIC comm interface



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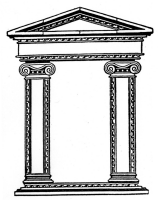
Package size: 15x12x2cm  
Weight: 57g  
Serial/I2C comm interface



Platform architecture: hardware layer uses the open source ArduPilot Mega 2.5.2 (aka APM 2.5.2) board

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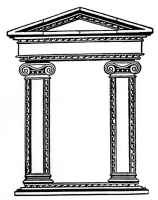
24bit AD converter  
SPI comm. interface



Platform architecture: hardware layer uses the open source ArduPilot Mega 2.5.2 (aka APM 2.5.2) board

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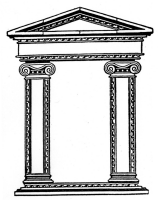
Max range: 7.65m



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Usmile Mini OSD MAVLink  
Flight Communications  
Module



Platform architecture: hardware layer uses the open source ArduPilot Mega 2.5.2 (aka APM 2.5.2) board

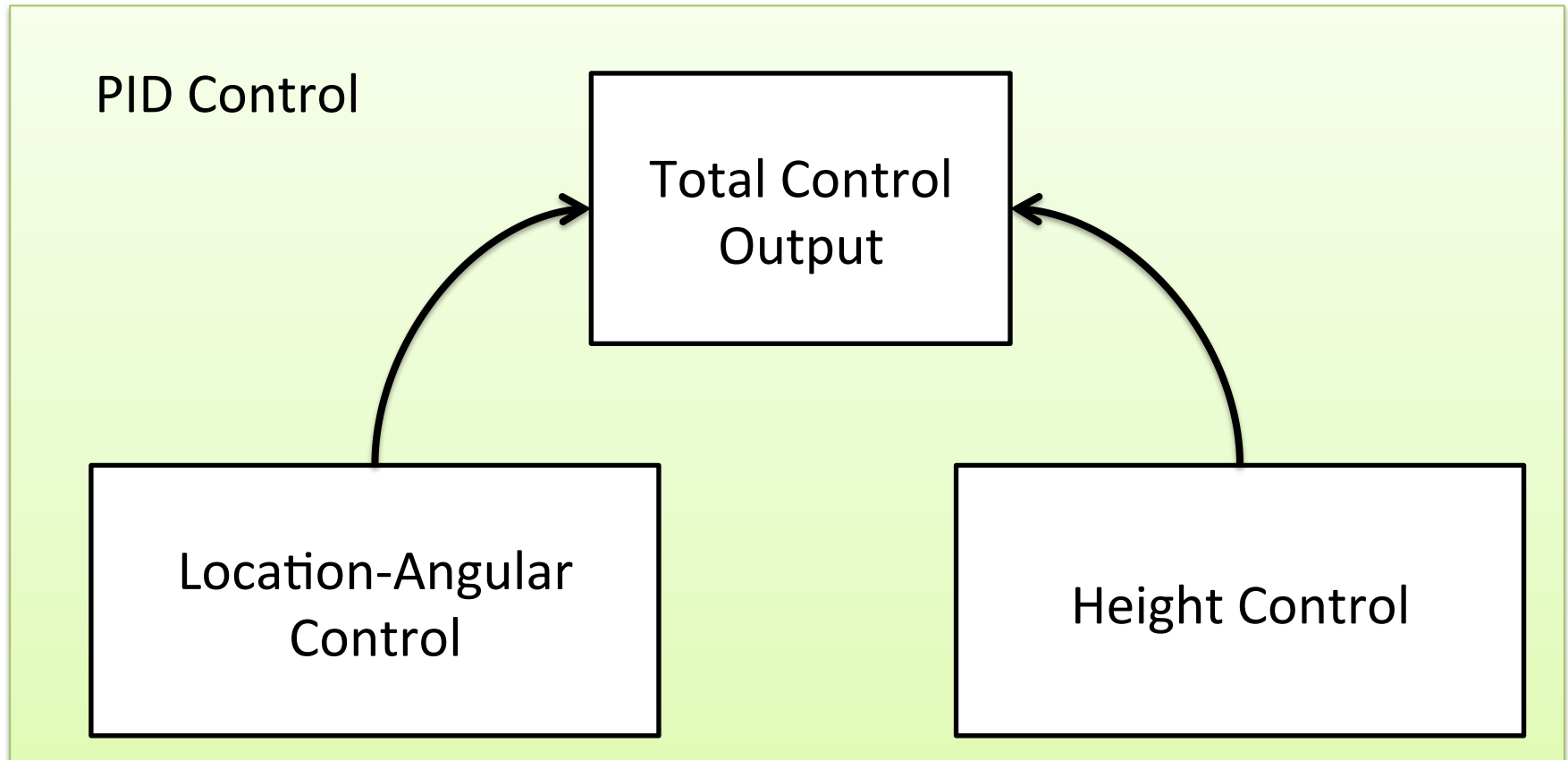
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Actuator:

Pulse Width Modulation (PWM) driven motors, duty cycle: 2.5msec (i.e. 400Hz)

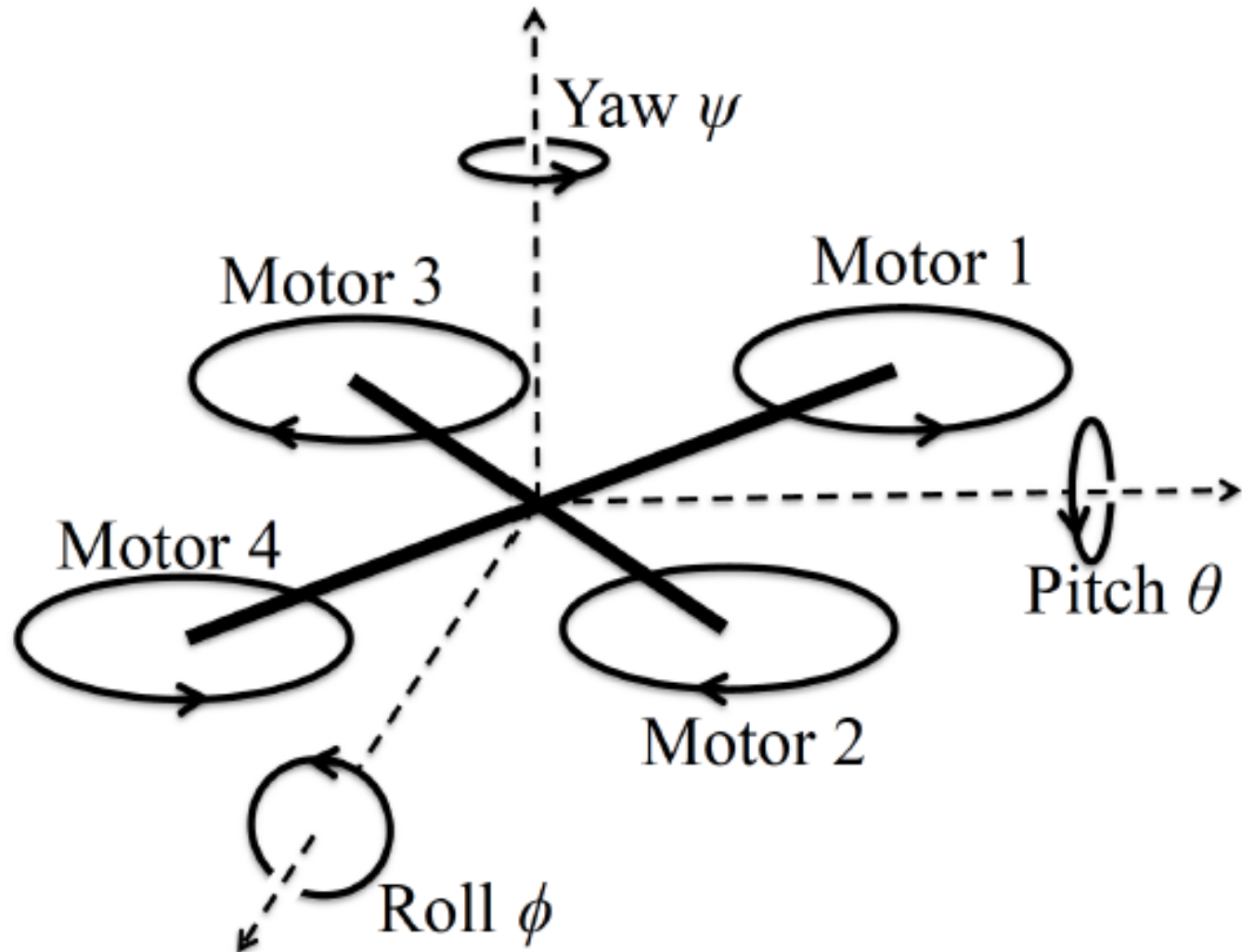


# Platform architecture: application layer implementation



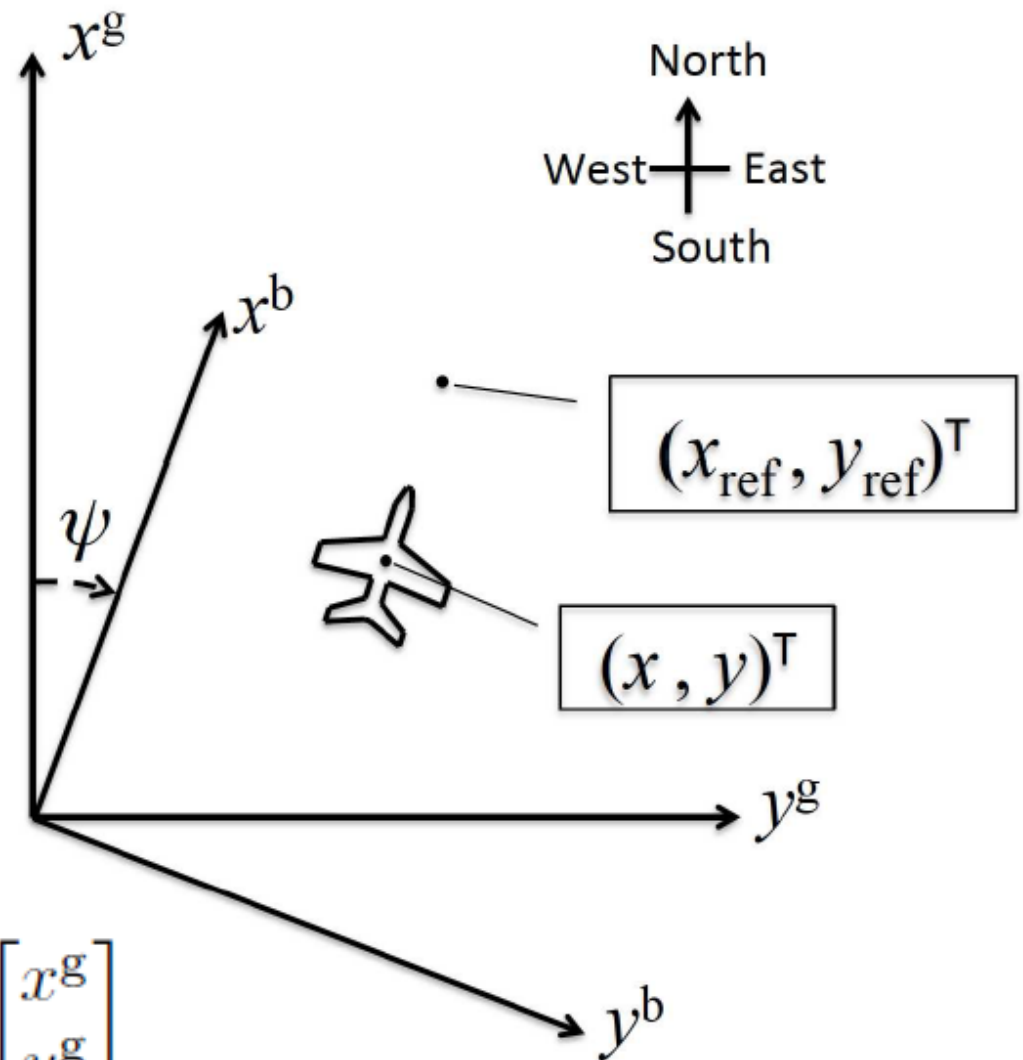


# Control Strategy: the quadcopter plant and states





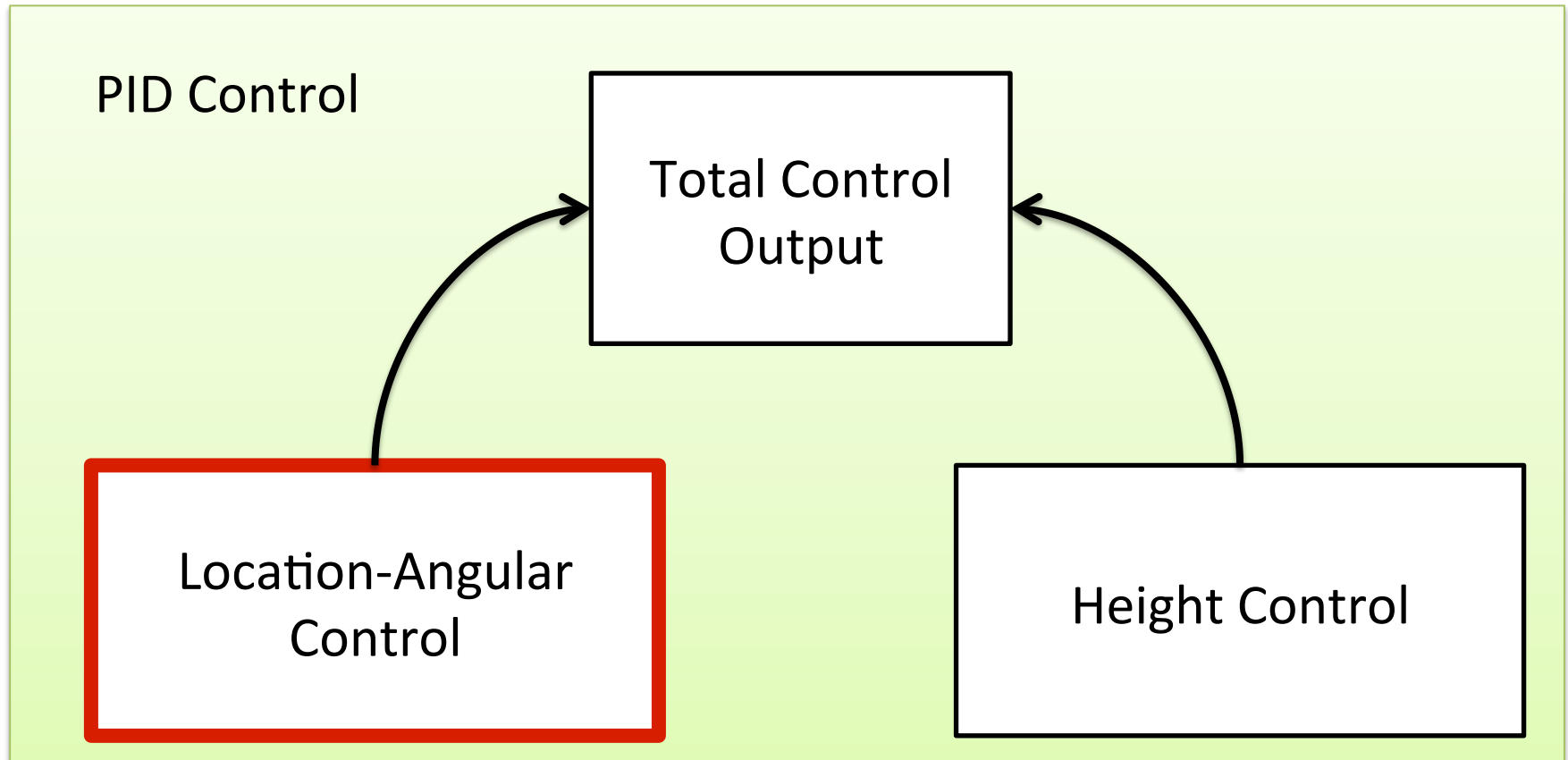
# Control Strategy: ground coordinates and body-oriented coordinates



$$\begin{bmatrix} x^b \\ y^b \end{bmatrix} = \begin{bmatrix} \sin\psi & \cos\psi \\ \cos\psi & -\sin\psi \end{bmatrix} \begin{bmatrix} x^g \\ y^g \end{bmatrix}$$



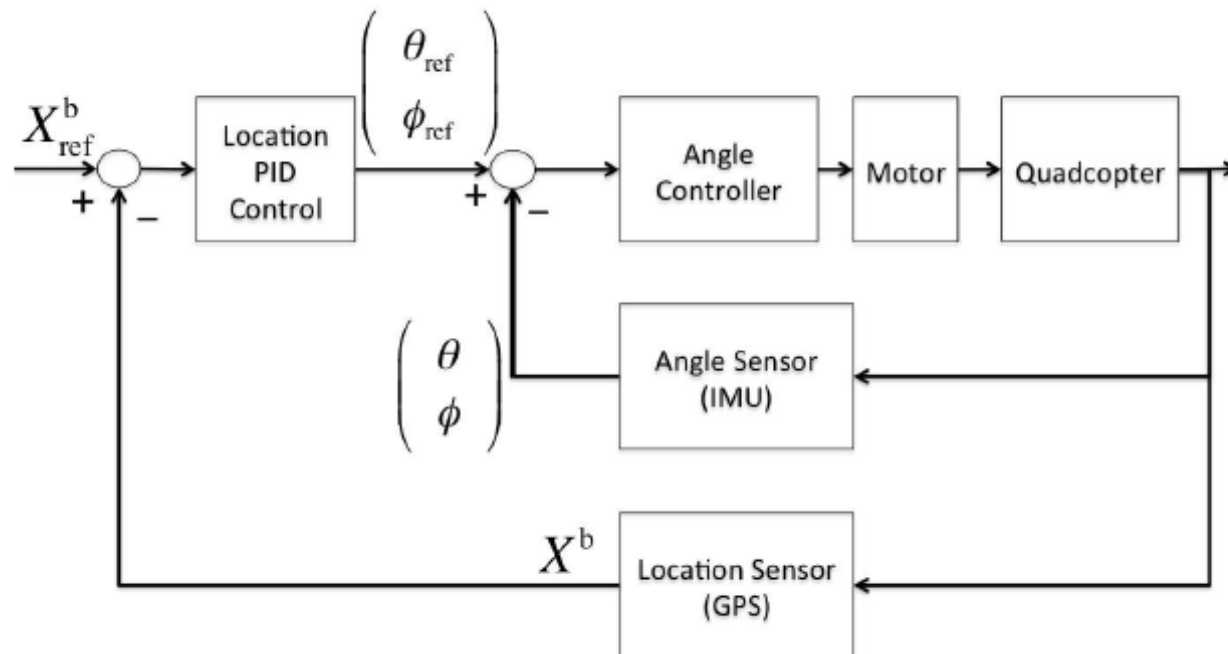
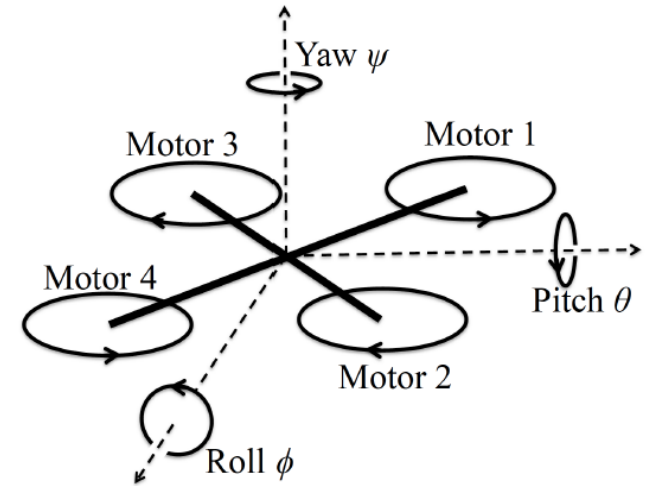
# Platform architecture: application layer implementation





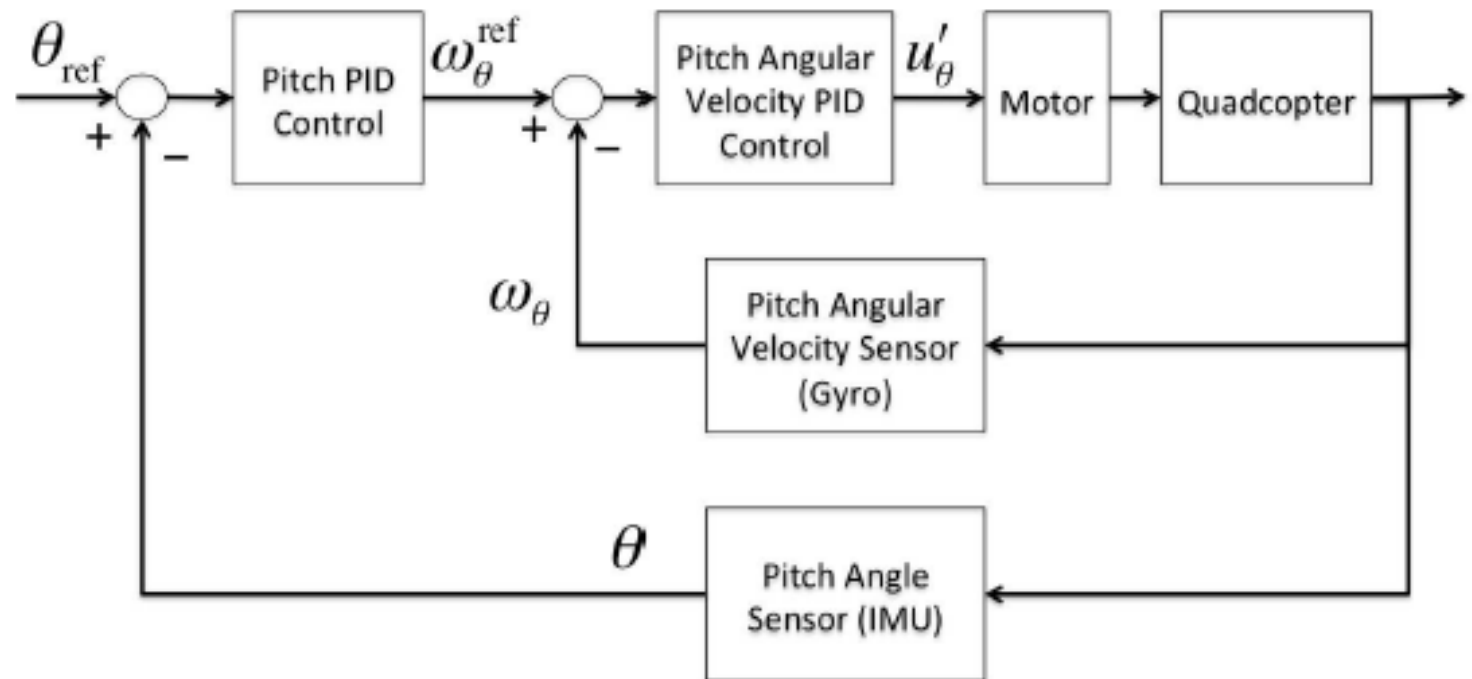
# location-angular control takes a nested outer-inner control loop form

$$\begin{aligned}\theta_{\text{ref}} &= k_x^p(x_{\text{ref}}^b - x^b) + k_x^d(\dot{x}_{\text{ref}}^b - \dot{x}^b) \\ &\quad + k_x^i \int_0^t (x_{\text{ref}}^b(\tau) - x^b(\tau)) d\tau, \\ \phi_{\text{ref}} &= k_y^p(y_{\text{ref}}^b - y^b) + k_y^d(\dot{y}_{\text{ref}}^b - \dot{y}^b) \\ &\quad + k_y^i \int_0^t (y_{\text{ref}}^b(\tau) - y^b(\tau)) d\tau,\end{aligned}$$

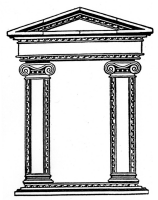




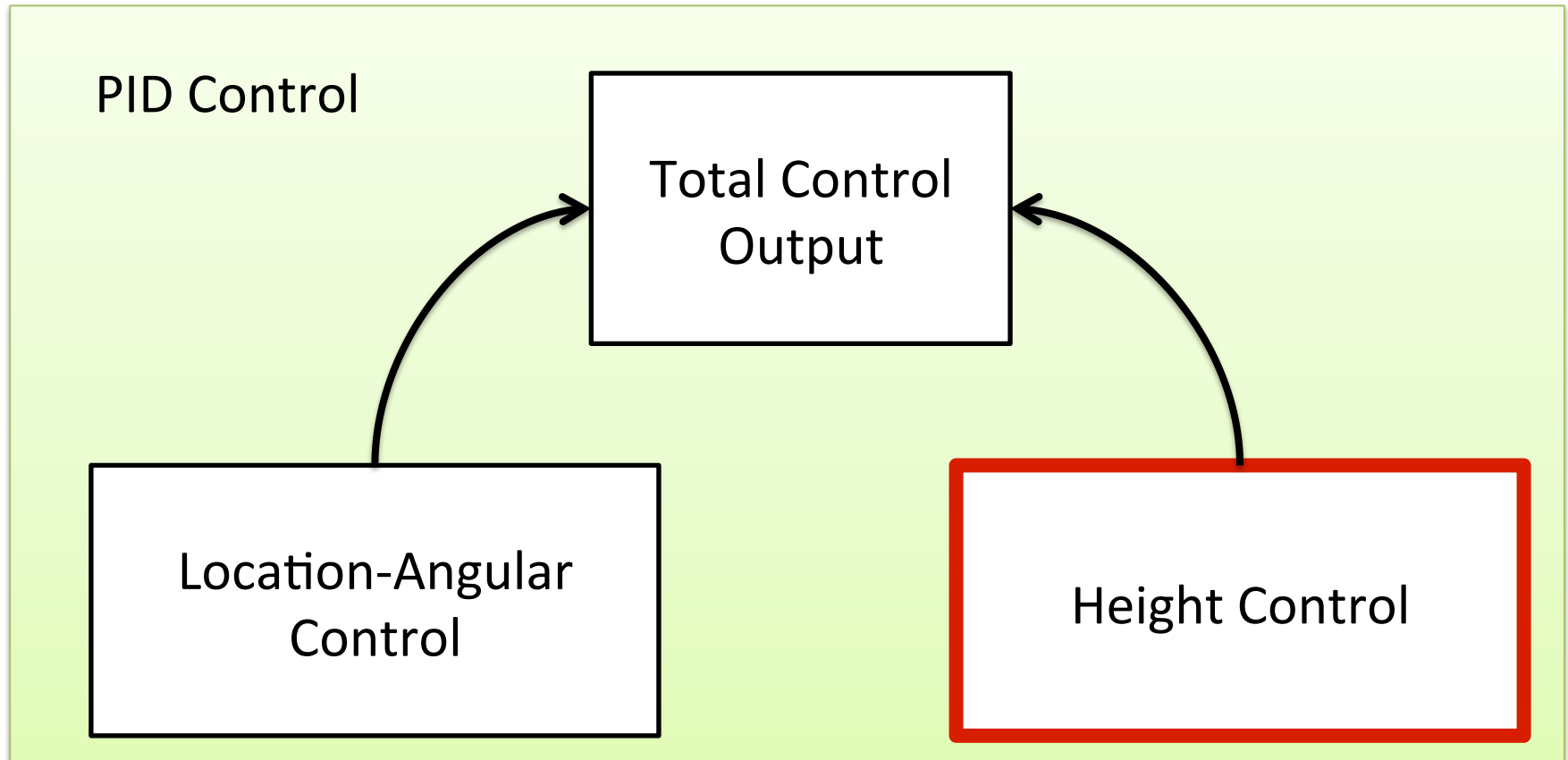
# Inner loop: accurate angular velocity readings, inaccurate angle readings



$$\begin{aligned}\omega_{\theta}^{\text{ref}} &= k_{\theta}^{\text{p}}(\theta_{\text{ref}} - \theta) + k_{\theta}^{\text{d}}(\dot{\theta}_{\text{ref}} - \dot{\theta}) \\ &\quad + k_{\theta}^{\text{i}} \int_0^t (\theta_{\text{ref}}(\tau) - \theta(\tau)) d\tau, \\ u'_{\theta} &= k_{\omega_{\theta}}^{\text{p}}(\omega_{\theta}^{\text{ref}} - \omega_{\theta}) + k_{\omega_{\theta}}^{\text{d}}(\dot{\omega}_{\theta}^{\text{ref}} - \dot{\omega}_{\theta}) \\ &\quad + k_{\omega_{\theta}}^{\text{i}} \int_0^t (\omega_{\theta}^{\text{ref}}(\tau) - \omega_{\theta}(\tau)) d\tau,\end{aligned}$$

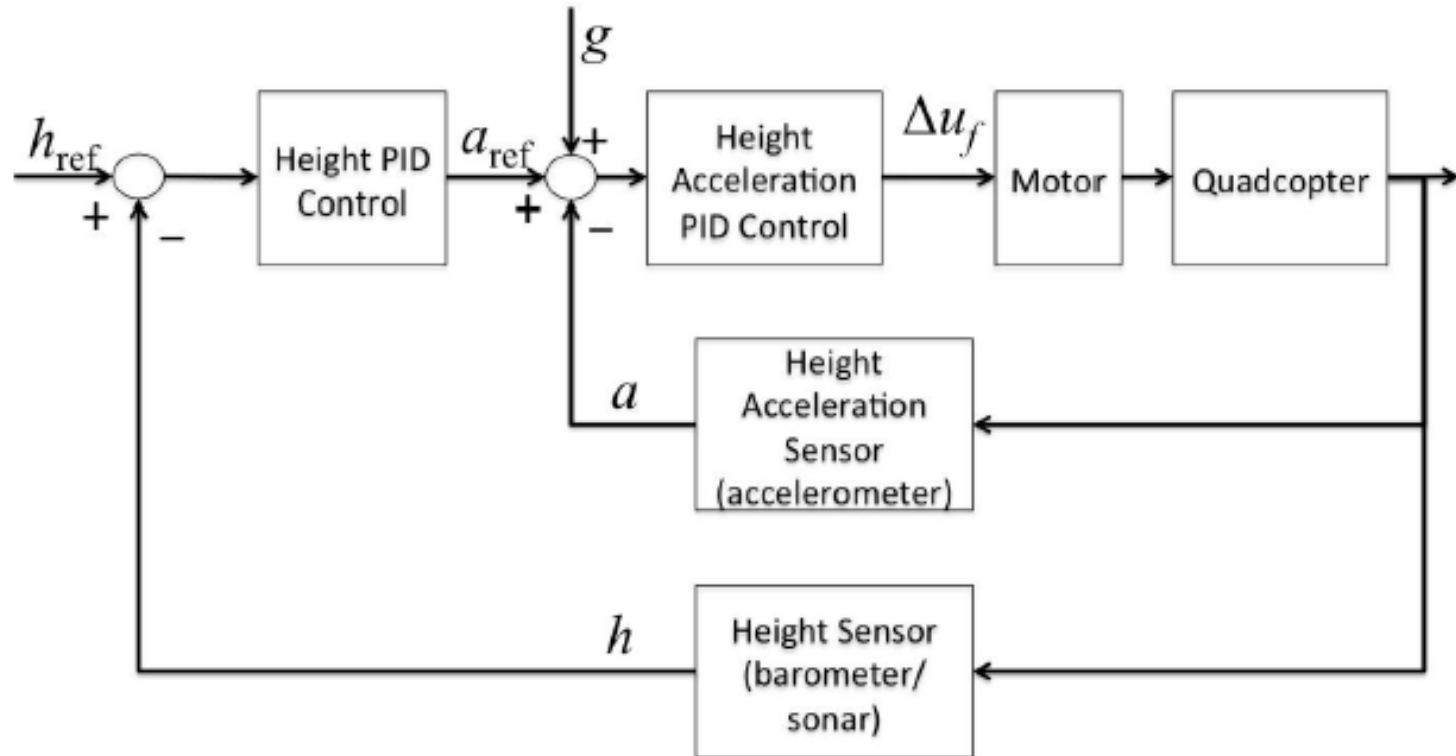


# Platform architecture: application layer implementation



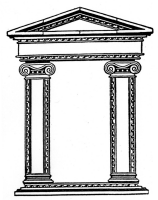


# Height control: accurate vertical acceleration readings, inaccurate height readings

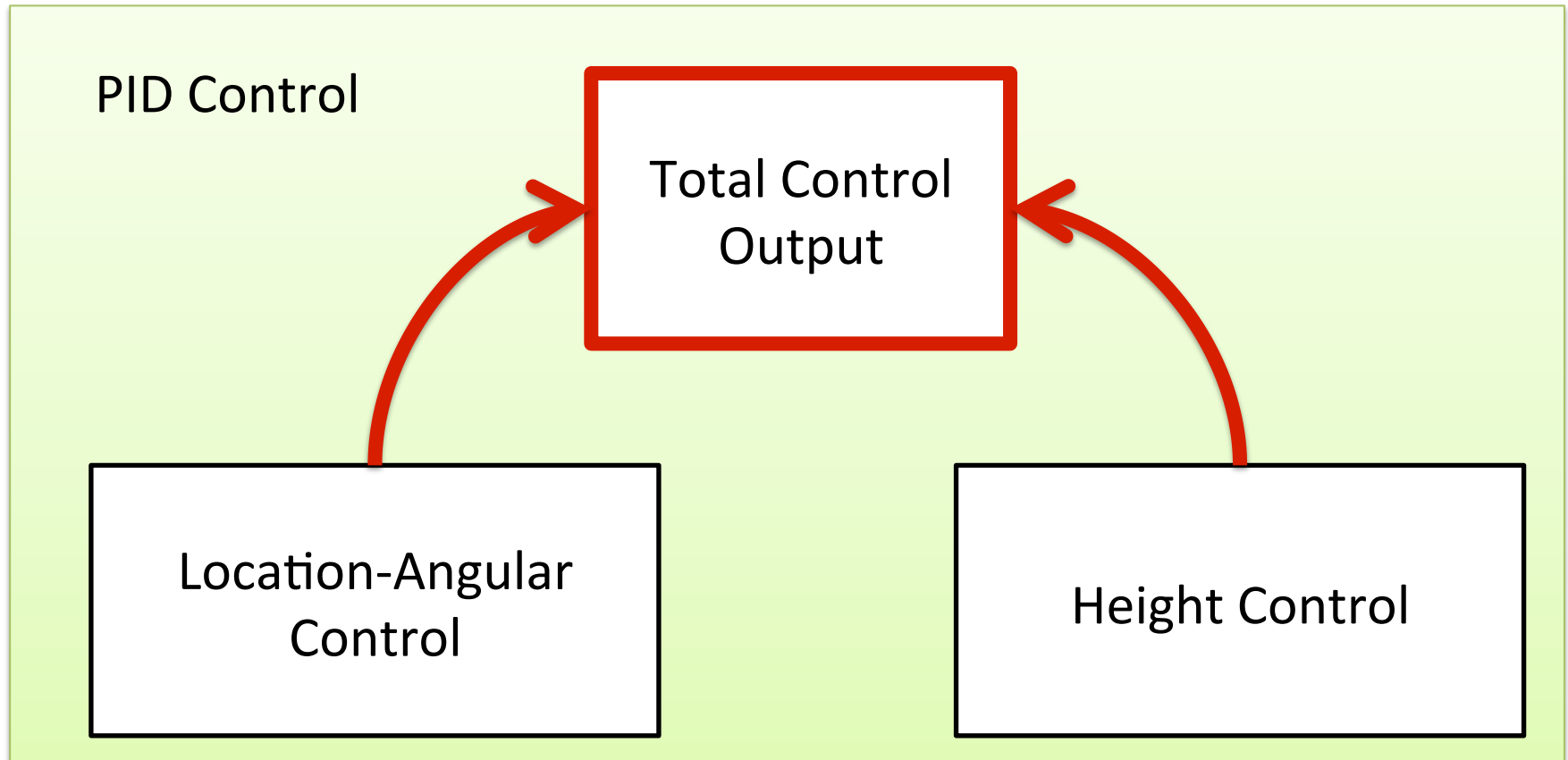


$$a_{\text{ref}} = k_h^p(h_{\text{ref}} - h) + k_h^d(\dot{h}_{\text{ref}} - \dot{h}) + k_h^i \int_0^t (h_{\text{ref}}(\tau) - h(\tau)) d\tau,$$

$$\Delta u_f = k_a^p(a_{\text{ref}} + g - a) + k_a^d(\dot{a}_{\text{ref}} + \dot{g} - \dot{a}) + k_a^i \int_0^t (a_{\text{ref}}(\tau) + g(\tau) - a(\tau)) d\tau,$$



# Platform architecture: application layer implementation





All above control outputs sum up to become the total control output to propeller motors

$$u_f(t + dt) = u_f(t) + \Delta u_f(t)$$

Updated throttle  
component

Height control  
output



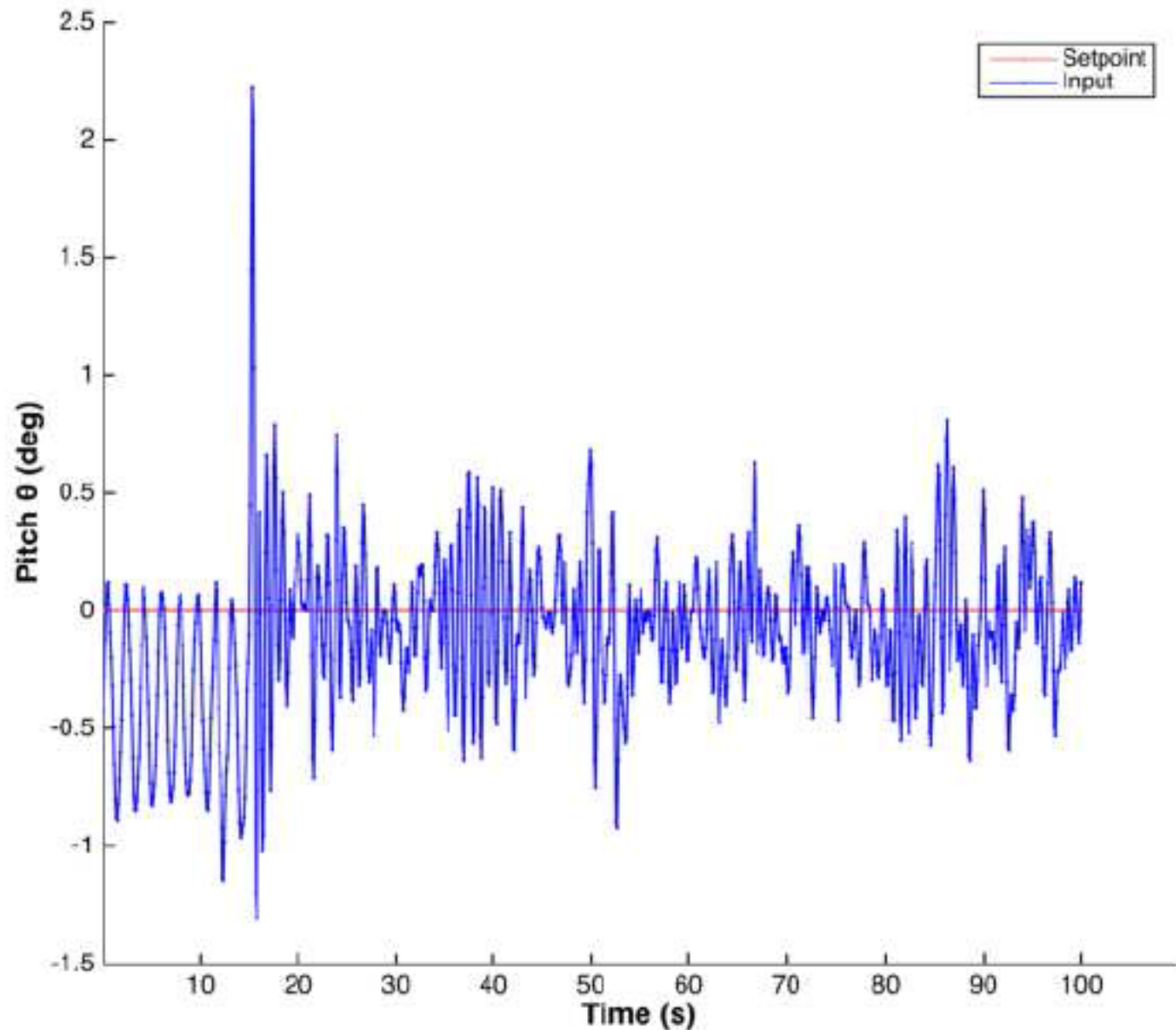
All above control outputs sum up to become the total control output to propeller motors

$$u_f(t + dt) = u_f(t) + \Delta u_f(t)$$

$$\begin{aligned} U_1(t + dt) &= u_f(t + dt) + u'_\theta(t) + u'_\phi(t) - u'_\psi(t), \\ U_2(t + dt) &= u_f(t + dt) - u'_\theta(t) + u'_\phi(t) + u'_\psi(t), \\ U_3(t + dt) &= u_f(t + dt) + u'_\theta(t) - u'_\phi(t) + u'_\psi(t), \\ U_4(t + dt) &= u_f(t + dt) - u'_\theta(t) - u'_\phi(t) - u'_\psi(t). \end{aligned}$$

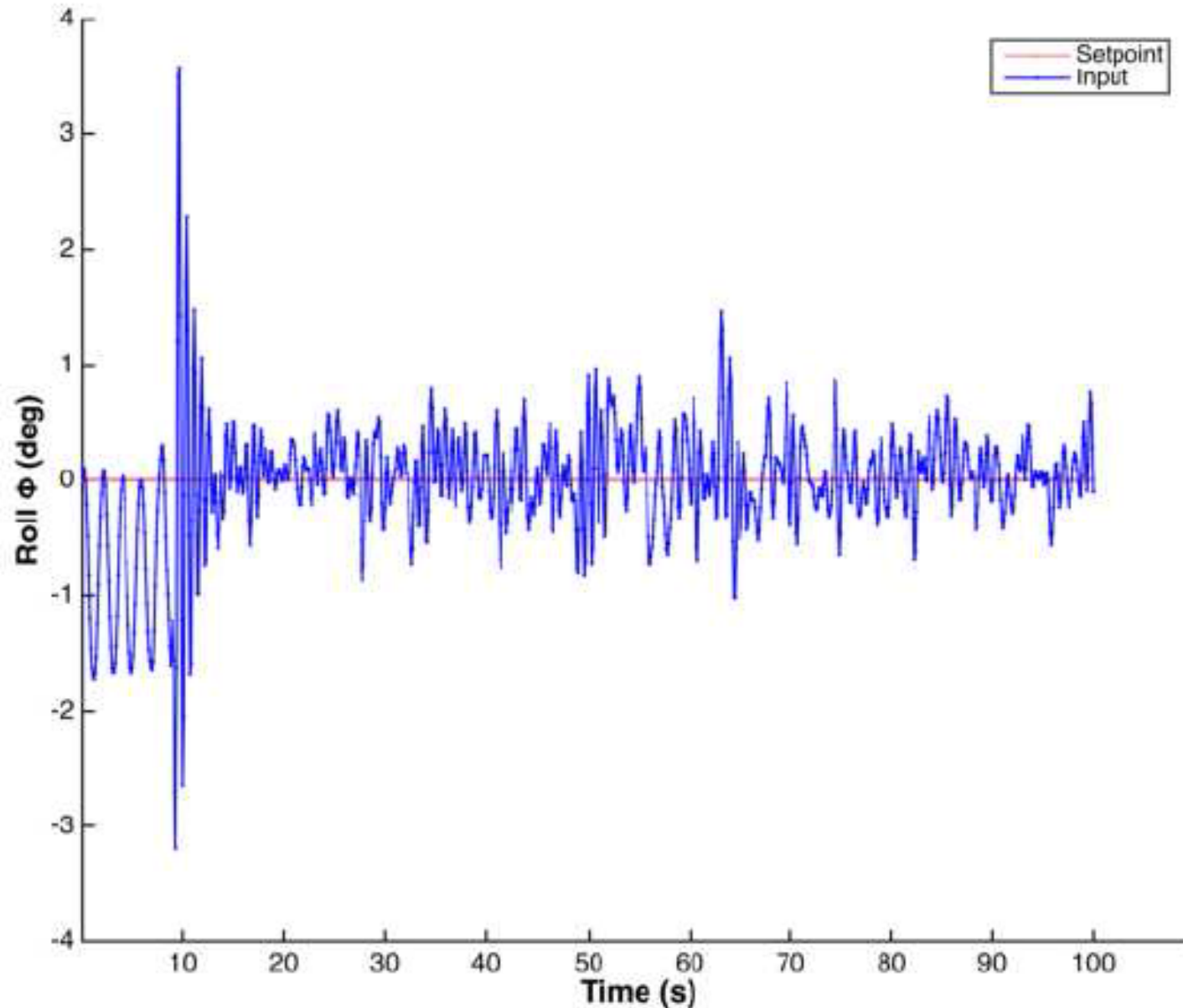


Pitch Angle Horizontal Stabilization (IMU starts at  $t = 0$ ; motors start at around  $t = 15$  sec; “Setpoint”: the desired pitch angle)



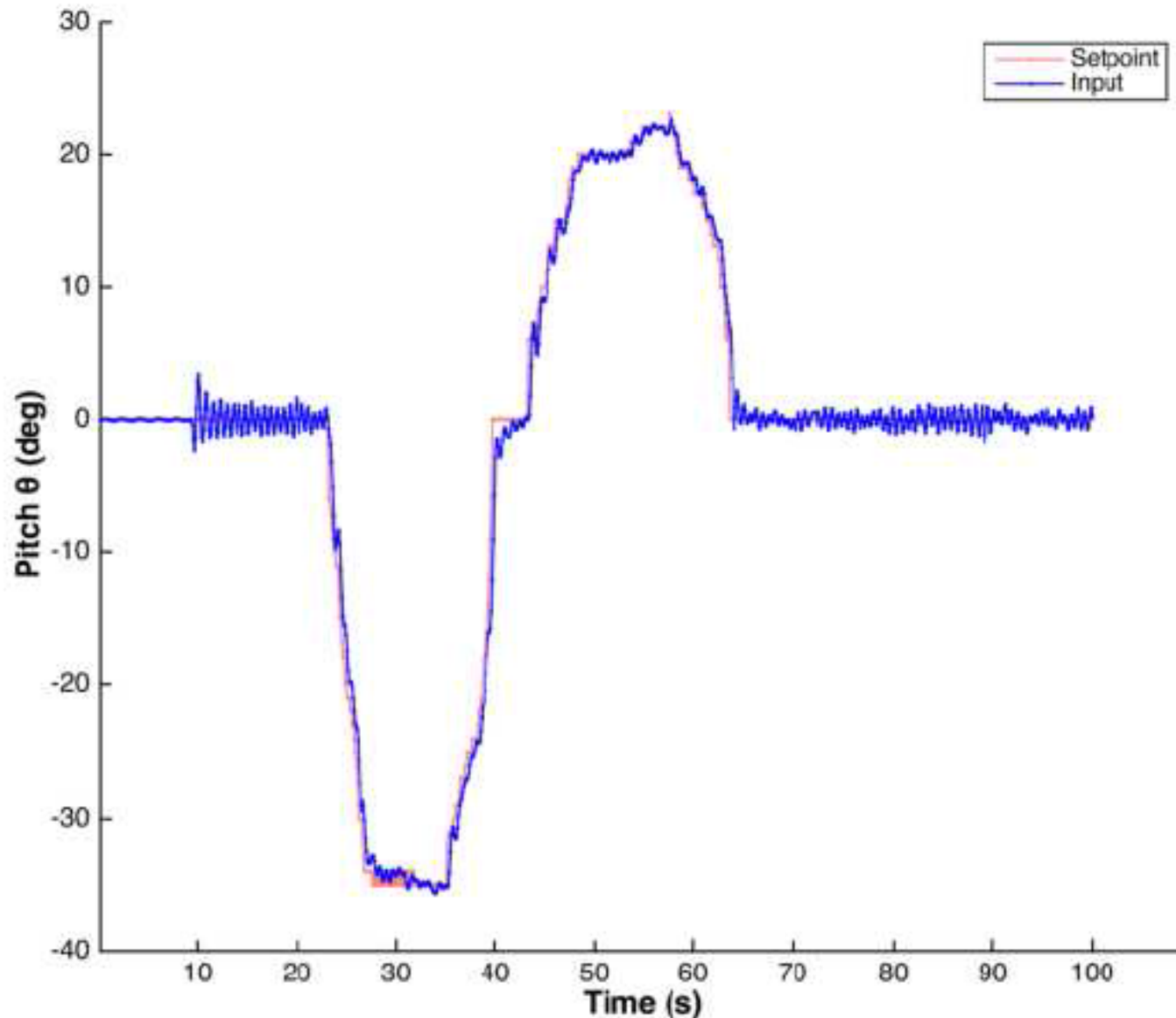


Roll Angle Horizontal Stabilization (IMU starts at  $t = 0$ ; motors start at around  $t = 9$  sec; “Setpoint” means the desired roll angle)



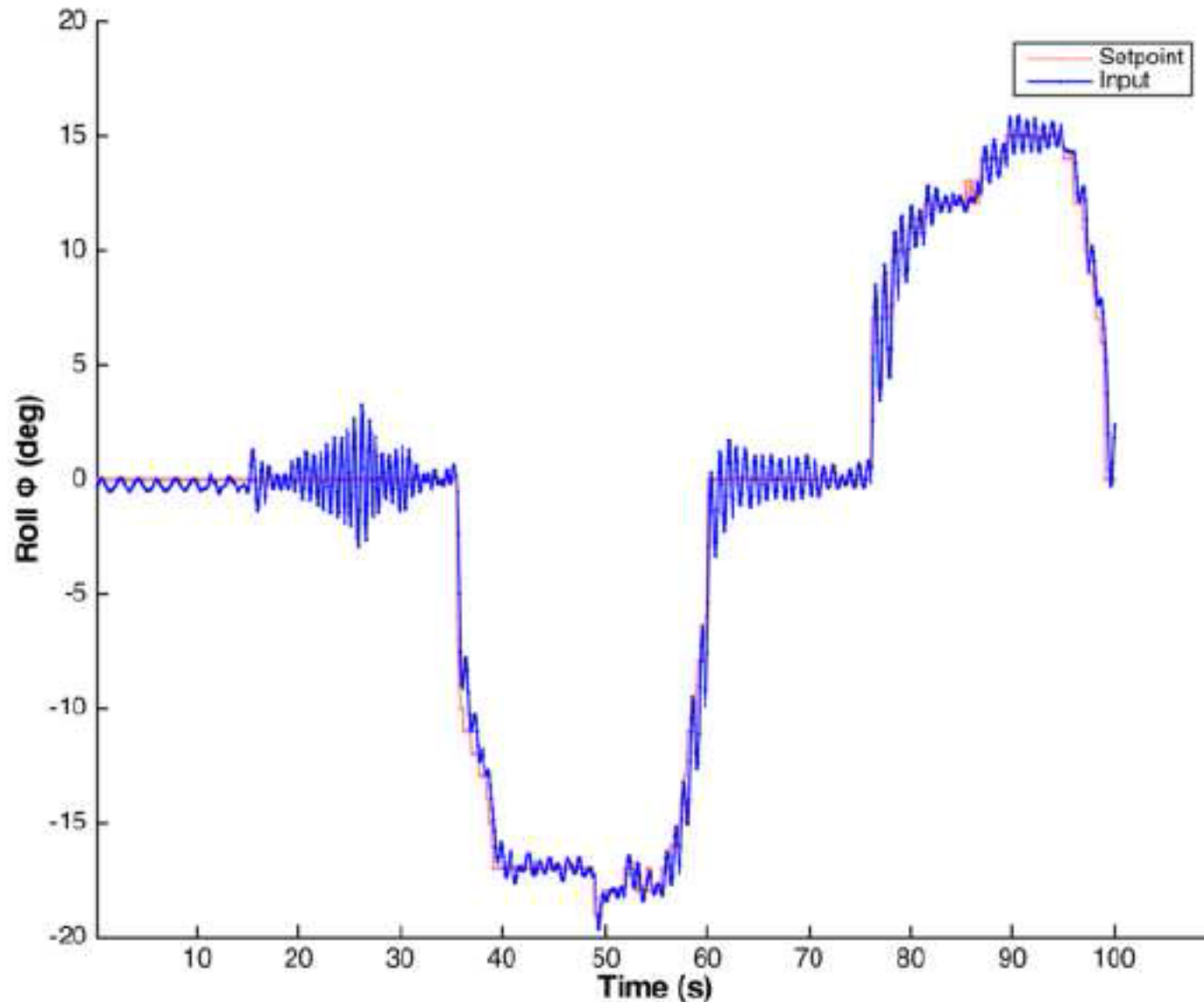


Pitch Tracking (IMU starts at  $t = 0$ ; motors start at around  $t = 10$  second; “Setpoint” means the desired pitch angle)



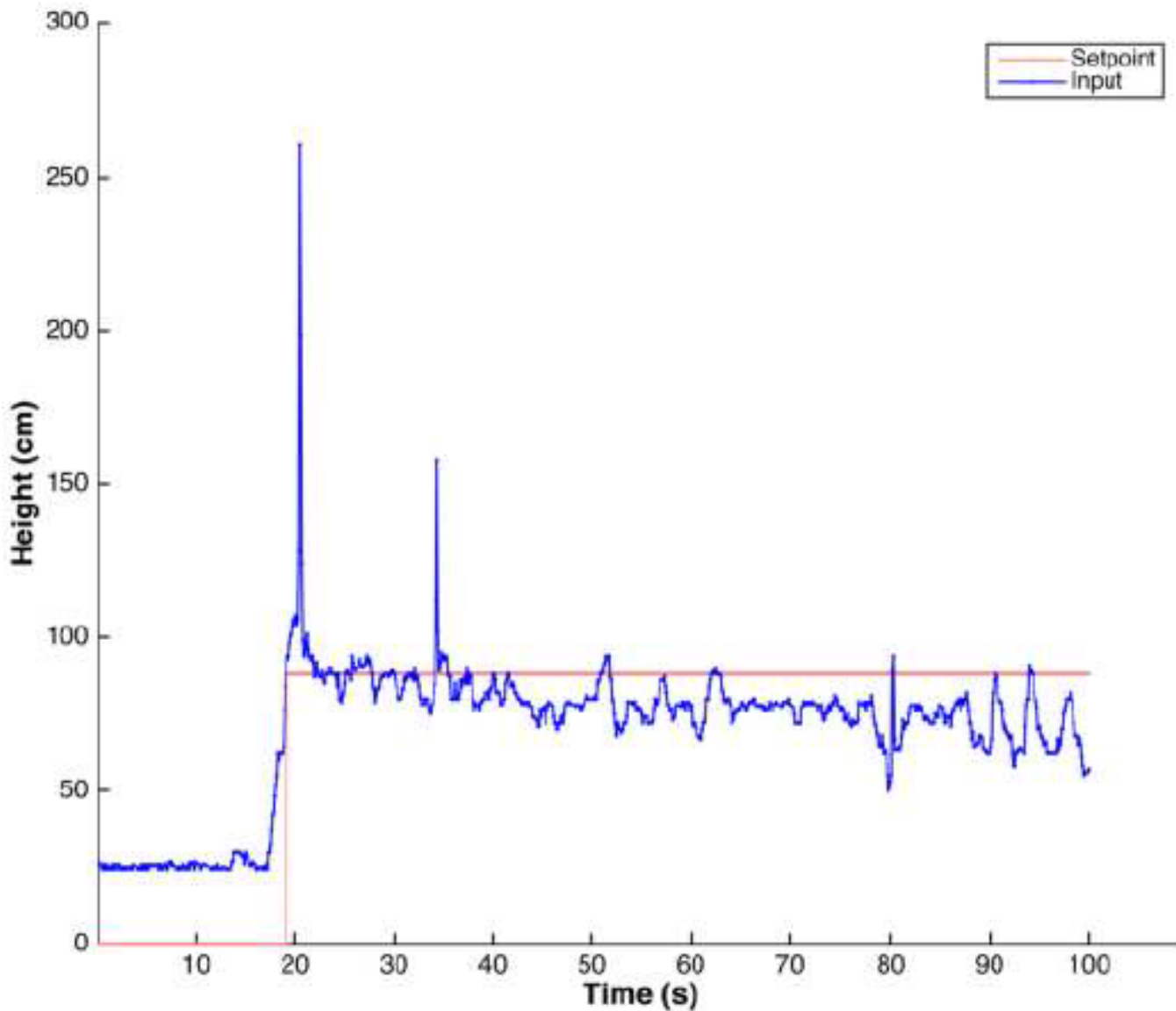


Roll Tracking (IMU starts at  $t = 0$ ; motors start at around  $t = 15$  sec; “Setpoint” means the desired roll angle)





Height Control Trace (IMU and sonar start at  $t = 0$ ; motors start at around  $t = 17$  sec; “Setpoint” means the desired height)





# Conclusion

We build an open source quadcopter platform.

Much simpler source code compared to  
Ardupilot (135KB application layer source code)

Good documentation



# Thank you!

We build an open source quadcopter platform.

Much simpler source code compared to  
Ardupilot (135KB application layer source code)

Good documentation



# Thank you!

Demo:

<http://www.comp.polyu.edu.hk/~csqwang/coolstuff>

Source Code URL:

<https://github.com/Ard-mu-copter/>